Experimental analysis of flow pattern and heat transfer in circular-orifice baffled tubes

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Abstract

An experimental study on the thermal-hydraulic and flow pattern characteristics of tubes with circular-orifice baffled inserts is performed. A geometry with an orifice-to-tube diameter ratio of $d/D = 0.5$ and an interbaffle spacing equal to 1.5 D is tested in steady-state conditions. Isothermal friction factor tests in the range $10 < Re < 2200$ allow the laminar, transitional and turbulent flow regimes to be identified. Flow visualization by means of hydrogen bubbles is used to assess the main flow structures and their relation with the onset of transition, which occurs at $Re \approx 160$. Heat transfer experiments under uniform heat flux are conducted in order to obtain the Nusselt number as a function of Reynolds number, for $150 < Pr < 630$, using propylene-glycol as the test fluid. Numerical simulations are used to complement the visualization study and explain the role of the flow structures on the thermal-hydraulic behavior.

Keywords: heat transfer enhancement, turbulence promoters, transitional

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flow, circular-ring turbulator, baffled tube

Nomenclature

c_p	specific heat (J kg^{-1} K^{-1})
C_0	orifice coefficient (-)
d	orifice diameter (m)
D	tube inner diameter (m)
k	thermal conductivity (W m^{-1} K^{-1})
l	Tube length (m)
l_{rec}	recirculation length (m)
L	distance between consecutive baffles (m)
L_h	heated length (m)
L_p	distance between pressure ports (m)
$\dot{m}$	mass flow rate (kg/s)
N	number of tubes (-)
q	net heat transfer rate (W)
q''	net heat flux (W/m^2)
$R3$	performance evaluation criterion (-)
S	open area, $(d/D)^2$ (-)
t	Baffle thickness (m)
T	Temperature (°C)
u	mean velocity, based on the tube diameter, D (m)
$\dot{W}$	power consumption (W)
x	axial distance from the start of the heated area (m)
Δp	pressure drop (Pa)

Dimensionless groups

f	Fanning friction factor, $\frac{\Delta p}{2\rho u^2}\frac{D}{L}$
Gr	Grashof number, $g\beta q'' D^4/k\nu^2$
Nu	Nusselt number, hD/k
Pr	Prandtl number, $\mu c_p/k$
Re	Reynolds number, $\rho v D/\mu$

Greek symbols

β	coefficient of thermal expansion (K^{-1})
μ	dynamic viscosity (kg m^{-1} s^{-1})
ν	kinematic viscosity (Pas)
ρ	fluid density (kg m^{-3})
σ	standard deviation

Subscripts

b	bulk
in	inlet
j	section number
s	smooth tube
wi	inner wall

1. Introduction

Insert devices are widely used means of heat transfer enhancement, which can be installed in smooth tubes of heat exchangers while maintaining their original mechanical strength. Likewise, retrofitting of existing equipment with tube inserts is also a remarkable advantage of this type of devices. Twisted tapes [1, 2, 3], wire coils [4, 5] and wire meshes [6] are typical designs used as insert devices in heat exchangers. Their geometrical characteristics must be chosen according to the operational conditions of the internal flow, in order to maximize convective heat transfer enhancement, keep pressure drop under reasonable levels and -if applicable- achieve turbulent promotion at low Reynolds number. Equally-spaced circular rings are another typology of insert device, where convective heat transfer is augmented on the basis of the periodic contraction and expansion of the bulk flow. They consist of flat disks with a central orifice, whose open area fraction is typically characterized by the orifice-to-tube diameter ratio d/D.

A limited number of experimental studies have analyzed the thermal-hydraulic performance of circular rings. Kongkaitpaiboon et al. [7] studied the heat transfer and pressure drop for circular-ring turbulators, testing several diameter ratios ($d/D = 0.5$, 0.6 and 0.7) and three pitch ratios, L/D (free space between baffles divided by the tube inner diameter), achieving a maximum thermal performance of 7% for the lowest pitch ratio ($L/D = 6$) and the highest d/D value. Promvonge et al. [8] tested inclined annular baffles, varying the relation between the disk thickness and the tube diameter, and the pitch ratio (0.5, 1.0, 1.5, 2.0) for an angle of 30°. A maximum value of 40% was found for the thermal performance for the lowest thickness/diameter ratio and the lowest pitch ratio. Acir et al. [9] studied heat transfer in tubes with annular disks, focusing on the effect of the pitch ratio L/D, and the number of orifices over the annulus; a maximum thermal performance of 83% was found for the geometry with two orifices and $L/D = 2$. Ruengpayungsak et al. [10] modified the annular geometry to include some semi-circular orifices to achieve a gear-ring geometry; a maximum thermal performance for $L/D = 3$ of 24%, 26%, 28% and 30% was found for baffles with tooth numbers of 0, 8, 16 and 24. In addition, circular-rings have been used in combination with other geometries as twisted tapes. Eiamsa-ard et al. [11] studied the thermal performance of a tube with twisted-tape and circular ring inserts. Several twist ratios (3.0, 4.0 and 5.0) and pitch ratios (1.0, 1.5 and 2.0) were tested. The maximum thermal performance, 42%, was achieved at the lowest twist and pitch ratios and Reynolds number tested ($Re \approx 6000$). Abolarin et al. [12] tested u-cut twisted tapes with ring inserts, obtaining earlier transitions and an increase in the pressure drop and the heat transfer when the rings were inserted, and the distance between them was lower.

From these studies, it can be concluded that these inserts show a high thermal performance when their geometry is well selected. It is also remarkable that all of them detected the highest thermal performance at the lowest Reynolds number of their tests, which were in almost all cases above $Re = 3000$. However, many industrial applications that may require the use of insert devices for heat transfer enhancement purposes work in laminar or transitional flow regimes and high Prandtl number fluids. Furthermore, the effect of Prandtl number on heat transfer in tubes with circular rings is not reported in the open literature, as the majority of experiments only use air as test fluid.

In addition to the interest of circular rings as a passive technique, these inserts have spread their applications during the last decades, due to the potential benefit in enhancing mass and heat transfer when an oscillatory flow is superimposed to a net flow. Currently, this configuration is found in oscillatory-baffled reactors (OBR), where the inserts promote the continuous flow separation and reattachment. In OBRs, a high-residence time product flows across the circular tube, yielding very low net flow Reynolds numbers. A piston or bellow connected to the tube provides a superimposed oscillatory flow, which increases the radial mixing and promotes heat and mass transfer intensively [13]. A few number of experimental investigations have shed light on the heat transfer characteristics of OBRs with circular orifice baffles [14, 15, 16]. However, the performance of the tube insert in the absence of oscillatory flow is still insufficiently described, correlations for Nusselt number are available only for the turbulent region (Reynolds numbers above 200) and the role of the Prandtl number has not been studied properly. In addition, González-Juárez et al. [17] studied numerically the heat transfer in baffled tubes and reported unrealistic behaviors for the proposed correlations [14, 15] after extrapolating at lower Reynolds numbers, because these provided lower heat transfer rates than a smooth tube. The need to increase the studied range of Reynolds numbers is justified by some potential applications, where the required residence time could be higher, or the fluid has a very high viscosity.

The experimental analysis of the flow pattern in circular rings has also attracted little attention up to date. Kiml et al. [18] performed heat transfer and visualization tests in tubes with transverse and inclined circular rings, with $d/D = 0.8$, working in the Reynolds number range $[5000 - 20000]$. Focusing on the transverse ribs, they identified the separation bubbles just behind the ribs and the reattachment point between consecutive ribs. This flow pattern showed a clear influence on the local Nusselt number results; after the flow separation downstream the rib, a sharp drop in heat transfer was noticed past the reattachment point due to the boundary layer growth. Muñoz et al. [19] used the Particle Image Velocimetry technique to study the behavior of the flow in equally-spaced baffles in round tubes with $d/D = 0.5$, $L/D = 1.5$ and $20 < Re < 300$. The measurement of the turbulent intensity and the energy of the fluctuating components of the flow field allowed to assess the unstable nature of the flow for $Re > 160$.

Experimental correlations for Fanning friction factor in the low Reynolds number range (e.g. $Re < 3000$) have not been reported so far in the open literature. An attempt to quantify the steady-state pressure drop in net and oscillatory superimposed flows was accounted for in the quasy-steady model proposed by Jealous and Johnson [20]. This model assumes that the pressure drop caused by a steady-state flow along a tube with equally-spaced annular baffles is the same as a series of individual orifices in a turbulent flow, using the discharge coefficient concept. This approach presents two main drawbacks: a) the discharge coefficient is strongly influenced by the flow regime [21], particularly in the laminar and transitional ranges, which advises against the use of a constant discharge coefficient in some applications; b) the interaction of the flow between consecutive baffles is not accounted for if the discharge coefficient is used for evaluating the overall pressure drop.

The present study aims at clarifying some open questions on the thermal-hydraulic characteristics and flow mechanisms in tubes with equally-spaced annular baffles working in the laminar and transitional flow regimes. A baffle geometry with $d/D = 0.5$ and pitch ratio $L/D = 1.5$ is used as test specimen, following the widely accepted designs for the construction of oscillatory baffled reactors proposed by Ni et al. [22]. Heat transfer tests under uniform heat flux conditions are conducted for a range of net-flow Reynolds number $10 < Re < 3000$, using propylene-glycol as working fluid at different temperatures, which allows to extend the Prandtl number analysis to the range $150 < Pr < 630$. Friction factor measurements obtained under isothermal conditions are presented in the range $10 < Re < 3000$, allowing the transition from laminar to turbulent flow to be identified. The physics of transition are described experimentally using hydrogen bubble visualization, and CFD results are also employed to support the discussion of the impact of the flow structures on the thermohydraulic behaviour.

2. Experimental test rig

2.1. Geometry

The geometry under study is shown at Fig. 1. The insert baffles are a geometrical annuli fitted to the inner tube wall, with an orifice diameter $d = 0.5D$. The baffle spacing is $1.5D$. The baffles are made of PEEK plastic (to avoid electrical conduction from the tube wall) with thickness $t = 1mm$. They are assembled using three axial rods (1.6 mm diameter).

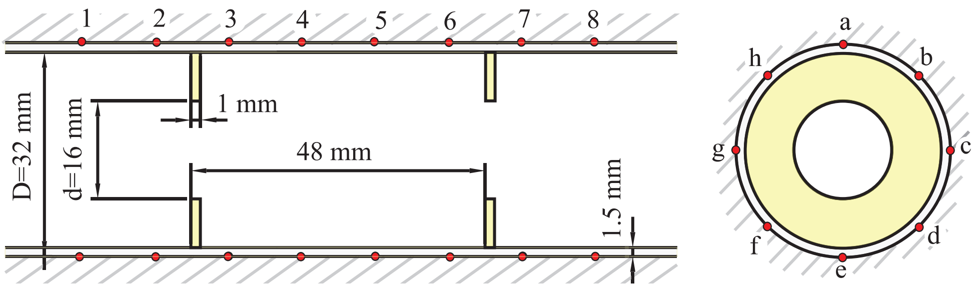


Figure 1: Baffle geometry and thermocouple arrangement in the test section.

2.2. Visualization facility

The facility depicted in Fig. 2 was built in order to study the flow within the described geometry by using hydrogen bubble visualization.

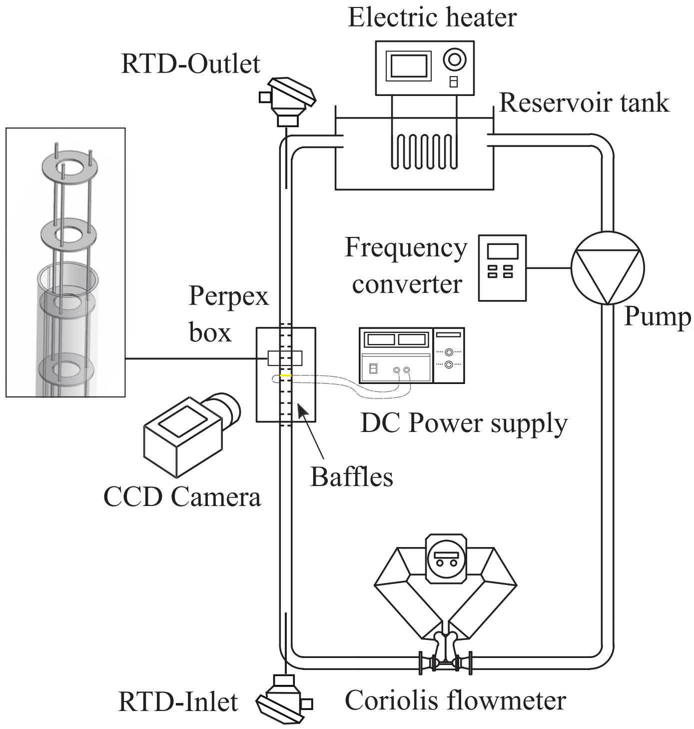


Figure 2: Experimental setup for hydrogen bubble visualization.

The test section consists of a $D = 32$ mm diameter acrylic tube (also depicted at Fig. 2) equally spaced insert baffles which are fixed by three steel rods. The use of the rods has been avoided in the visualization section for better results. Five baffles are placed upstream of the test section and also downstream of it in order to ensure spatial periodicity of the flow field. A flat-sided acrylic box, filled with the same working fluid used in the facility, is placed around the test section for a better optical access.

Heating and final temperature control is carried out by an electric heater located in the upper reservoir tank. A Coriolis flowmeter and a control valve are used to control and monitor the working flow rate.

The visualization by hydrogen bubbles is a qualitative technique that allows the perception of the main flow structures. For that, the hydrogen bubbles are generated by means of a copper wire (cathode) that crosses the tube diametrically upstream of the visualization area, being the anode a metallic component of the circuit located downstream of the test section. Thus, a symmetry plane of the flow is seeded with hydrogen bubbles, while illumination is provided by two rear lamps. A CMOS 1280x1024 pix^2 CMOS camera is situated in orthogonal position in relation to that plane, so that it can have a front view of it, or in parallel to it in order to have a side view. The bubbles size and quantity are set up by adjusting the direct current voltage applied between cathode and anode. Finally, In order to work with feed voltages below 50 V, the electrical conductivity of the working fluid is increased by adding salt.

2.3. Thermal-hydraulic tests

A schematic diagram of the experimental set-up is shown in Fig. 3. It consists of three independent circuits. The second and third circuits are used to regulate the temperature of the reservoir tank (1). Test fluid was pumped from the open reservoir tank (1) by a train of three variable-speed gear pumps (2). The flow rate was measured by a Coriolis flow-meter (3). The baffles are installed in the main circuit (5). The test section was a thin-walled, 2 m long, 316L stainless steel tube with 37 equally-spaced insert baffles. The inner and outer diameters of the tube were 32 mm and 35 mm, respectively

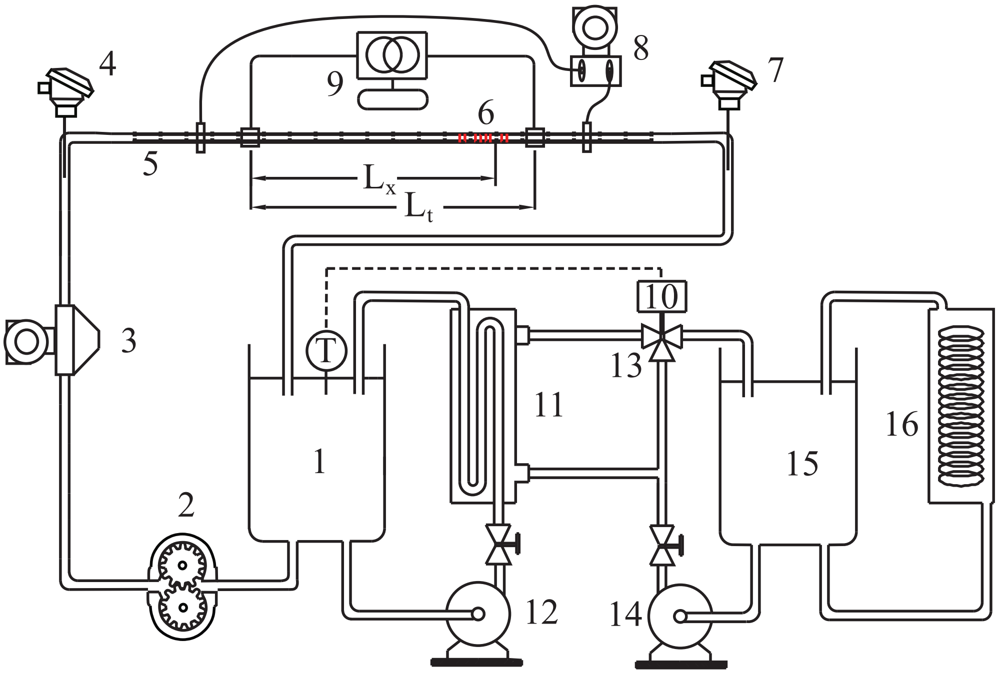


Figure 3: Experimental set-up. (1) Reservoir tank, (2) Pumping system, (3) Coriolis Flowmeter, (4) PT-100 Class B 1/10 DIN temperature sensors, Inlet, (5) Baffles, (6) Wall thermocouples, (7) PT100, Outlet, (8) Pressure transducer, (9) Autotransformer, (10) PID Controller, (11) Heat Exchanger, (12) Centrifugal Pump, (13) Three way valve, (14) Centrifugal Pump, (15) Reservoir tank, (16) Chiller.

Heat transfer experiments were carried out under uniform heat flux (UHF) conditions. The tube was heated by Joule effect through AC in the tube wall. Power was supplied by a 6 kVA transformer (9) connected with copper electrodes to the tube. A variable auto-transformer was used for power regulation. The loop was insulated by an elastomeric thermal insulation material to minimize heat losses. The overall electrical power added to the heating section was calculated by measuring the voltage between electrodes and the electrical current. Fluid inlet (4) and outlet (7) temperatures were measured by submerged type RTDs (Resistance Temperature Detector).

Fig. 1 shows the wall temperature measurement lay-out installed along the test section, using T-type thermocouples. The measurement section is located 13 cell tanks downstream of the first electrode, to ensure that the flow is thermally developed (spatially-periodic). A total number of 64 thermocouples peripherally distributed along eight axial sections on the outside wall cover two consecutive mixing tanks, following the sketches of Fig. 1. This arrangement is aimed at detecting the circumferential temperature gradient due to the flow stratification in the laminar region, and the axial temperature variations due to the local flow characteristics that occur between consecutive baffles. Pressure drop tests were carried out under isothermal conditions. In order to capture the characteristics of the spatially-periodic flow, the first pressure port is placed in the fifth inter-baffle spacing. The second pressure port is located in the twenty-seventh inter-baffle spacing, which falls at a distance $L_p = 1.296$ downstream. Four pressure holes separated by 90° are made in the pressure connections in order to accommodate any peripheral disturbances of the static pressure. A set of highly accurate capacitive differential pressure transducers (range: 0-10 mbar, 0-50 mbar, 5-500 mbar, 22-2500 mbar) were employed to measure the pressure drop along the test section. Propylene-glycol is used as working fluid. To ensure the right characterization of the fluid viscosity, a calibrated Cannon-Fenske viscometer is used periodically. The fluid temperature in the main tank is regulated by two additional circuits, with a variation of 0.1°C of the target temperature.

2.4. Data reduction

Two different calibration tests were performed in order to conduct the heat transfer experiments appropriately. Firstly, calibration of the wall thermocouples was achieved by circulating very high flow rates of fluid at different temperatures. Steady-state measurements of the inlet and outlet temperatures and evaluation of the conduction thermal resistances across tube wall and insulation, allowed us to calibrate them, taking into account any contact resistance with the tube wall. Secondly, heat losses to the ambient were evaluated by measuring the fluid inlet and outlet temperatures when low flow rates were circulated through the test section at different temperatures. The circumferential average Nusselt number is calculated at each section, j, using the following expression:

$$\bar{N}u_j = \frac{q''}{\bar{T_w}i_{i,j} - T_{b,j}} \cdot \frac{D}{k} \tag{1}$$

where q'' is the generated heat (the voltage times the electric current) minus the heat losses along the test section, per heated area. $T_{w_{i,j}}$ is the averaged inner wall temperature at the j section, which is calculated using a two-dimensional numerical model which solves the radial and axial heat conduction equation in the tube wall. T_{b_j} is the bulk fluid temperature at the section j. Since the heat was added uniformly along the tube length, T_{b_j} was calculated by considering a linear variation with the axial direction, according to:

$$T_{b,j} = T_{b,in} + \frac{q}{\dot{m}c_p} \cdot \frac{x_j}{L_h} \tag{2}$$

where $T_{b,i}$ is the fluid temperature at the inlet of the test section and x_j is the axial distance between the test section j and the point where the tube heating starts. The circumferential average Nusselt number is corrected by the factor $(\mu_{w_i}/\mu_{w_b})^{0.14}$ to account for the change in the physical properties due to the temperature radial gradient [23]. Finally, the axial-averaged Nusselt number is calculated as:

$$\bar{Nu} = \frac{\sum_1^8 \bar{Nu}_j}{8} \tag{3}$$

Fanning friction factor was determined from the fluid mass flow rate and the pressure drop measurements as:

$$f = \frac{\rho\pi^2\, D^5\, \Delta p}{32\, \dot{m}^2\, L_p} \tag{4}$$

2.5. Uncertainty analysis

The experimental uncertainty was calculated by following the Guide to the expression of uncertainty in measurement' published by ISO [24]. Instrumentation errors are summarized in Table 1. Uncertainty calculations based on a 95% confidence level showed limit values of 5.4% for Reynolds number, 3.9% for Prandtl number, 6.3% for Fanning friction factor and 2.3% for Nusselt number.

3. Numerical model

A 3D model of a tube with eight equally-spaced baffles is created and meshed using structured, hexahedral grid. A double compression ratio was introduced in both sides of the baffles, where greater variations of the flow pattern

Measurement	Uncertainty
Bulk temperature	0.15 °C
Wall temperature	1.12 °C
Voltage	0.04% measure + 0.03% full scale
Intensity	0.1% measure + 0.04% full scale
Viscosity	3% measure
Pressure drop	0.2% measure
Thermal conductivity	0.9% measure
Specific heat	0.3% measure
Tube diameter	0.1% measure
Heat transfer section	0.01 m
Thermocouples position	0.005 m
Pressure test section	0.005 m

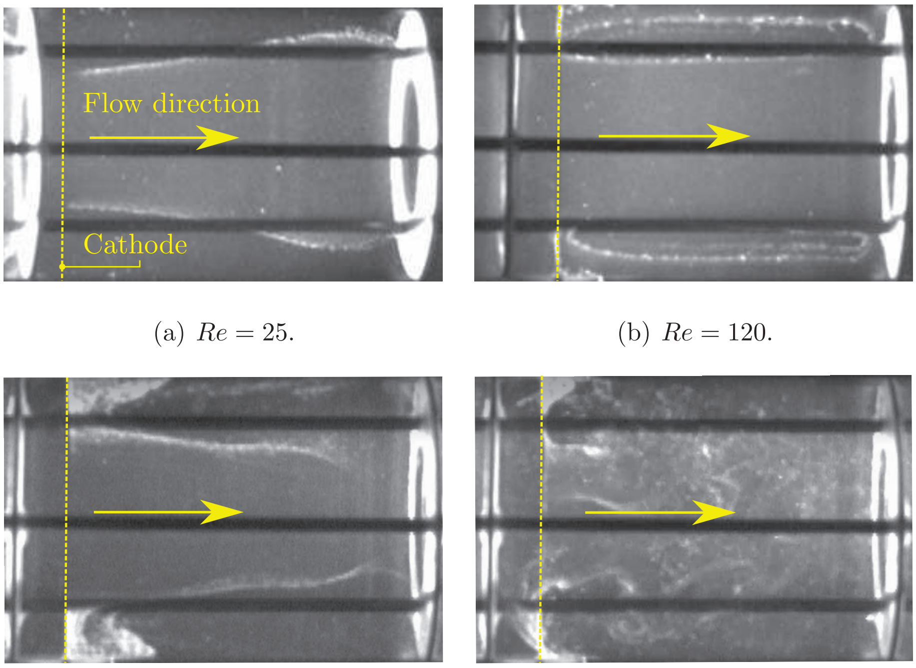
Table 1: Measurement uncertainties

were expected due to the geometry constriction, and in radial direction, to ensure better solution where higher-velocity gradients were expected. The finite volume software ANSYS Fluent 18 was employed for the solution of the continuity and momentum pressure-based equations. Full Navier-Stokes equations were treated in general, body fitted coordinates. A control-volume storage scheme was employed where all variables were stored at the cell center. A second order upwind scheme was used in order to interpolate the face values of computed variables. An implicit segregated solver solved the governing equations sequentially. In this study the pressure-velocity coupling algorithm SIMPLE was used. Steady-state simulations were solved in a first attempt, followed by the solution of the unsteady problem over a total time $t = 10$ s, using a time step $\Delta t = 0.1$ s, with a second-order discretization. This ensured a better convergence of the problem.

4. Flow visualization results

Experiments were carried out for Reynolds numbers between $25 < Re < 410$. The results are shown at Fig. 4 and Fig. 5 in front and lateral views.

In Fig. 4(a), the front view for $Re = 25$, a core jet can be observed downstream the baffle, which gets broader along the cell and then narrower again upstream the next baffle. Besides, low velocities are detected in the peripheral region. For $Re = 120$ the core jet has an uniform diameter along the interbaffle space, which is equal to the baffles orifice diameter, d. In Fig. 4(b), the bubbles generation was adjusted to show the recirculation of the outer region of the flow, which, again, has much lower velocities than the jet. By comparing the results for $Re = 25$ and $Re = 120$, the recirculation in the



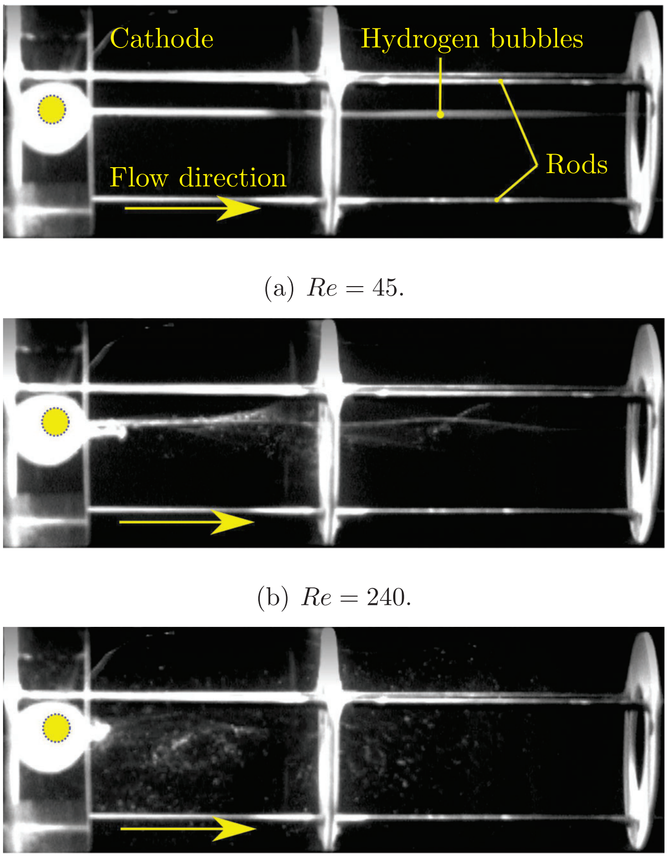
(a) $Re = 25$.

(b) $Re = 120$.

(c) $Re = 160$.

(d) $Re = 300$.

Figure 4: Front view of the flow seeded with hydrogen bubbles for different Reynolds numbers.



(a) $Re = 45$.

(b) $Re = 240$.

(c) $Re = 410$.

Figure 5: Lateral view of the flow seeded with hydrogen bubbles for different Reynolds numbers.

outer region of the flow is observed to grow with the Reynolds number. The results for this range of Reynolds numbers ($Re < 120$) show laminar flow conditions. This can be clearly detected from the lateral view of the flow depicted in Fig. 5(a), where the hydrogen bubbles remain in the same plane. At Reynolds number $Re = 160$ (Fig. 4(c)), a similar core jet dominates the flow, although significant pulsations are detected. Such pulsations are also observed, in a higher frequency, in the lateral view for $Re = 240$ in Fig. 5(b), where they result in part of the bubbles coming out of the symmetry plane. For higher Reynolds numbers the flow becomes chaotic. The front view of the flow field for $Re = 300$, depicted in Fig. 4(d), shows no appreciable pattern but a disordered flow, while the lateral view provides qualitative information about a high mass transfer taking place in radial direction. The same flow behaviour is observed in the experiment at $Re = 410$ in Fig. 5(c).

Text for electronic version only: The corresponding videos to the previous images can be also visualized. They show the front view of the flow seeded with hydrogen bubbles for: $Re = 25$ (Video 1), $Re = 120$ (Video 2), $Re = 160$ (Video 3) and $Re = 300$ (Video 4), and the lateral view for $Re = 45$ (Video 5), $Re = 240$ (Video 6) and $Re = 410$ (Video 7).

5. Numerical results

As could be observed in the previous section, the flow pattern seems to play a key role at low Reynolds numbers before the onset of the transition. The flow pattern is studied in more detailed in this subsection. So, several simulations have been performed in the laminar regime, from $Re = 1$ to $Re = 150$. The results, showing the velocity magnitude and the streamlines for a meridional plane along a cell (space between baffles), are represented in Fig. 6 for four Reynolds numbers.

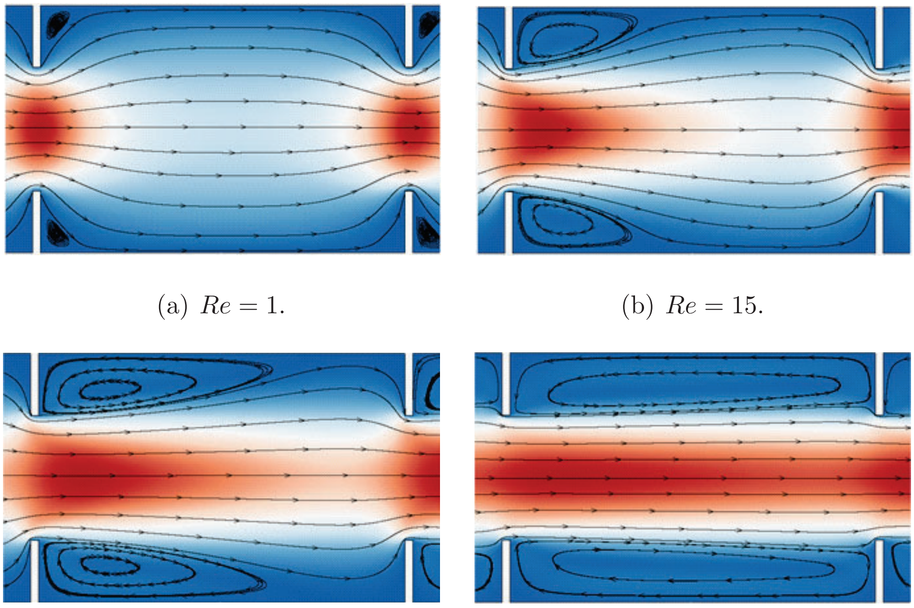
For the lowest Reynolds number simulated (Fig. 6 (a)), $Re = 1$, the core stream expands after flowing through the baffle growing up to the walls. Despite the low Reynolds number, a tiny recirculation can be already seen downstream the baffles. For Re=15, Fig. 6 (b), the core stream still reaches the wall after expanding, but the recirculation has grown substantially. A limit case can be seen at $Re = 32$, Fig. 6 (c), the core stream is not able to fully expand before reaching the next baffle. The recirculation fills more than one half of the interbaffle spacing. The extreme case corresponds to the pattern at $Re = 102$, Fig. 6 (d), the core stream expands very slightly, and the recirculation fills all the space between consecutive baffles, this way, the flow can be described as a short-circuit between baffles.

In order to quantify the evolution of the size of the recirculation area, the recirculation size is calculated by measuring the average distance from the baffle for which the axial velocity reverses. A non-dimensional definition of this size for different Reynolds number is presented in Fig. 7. The graph shows a growth of the recirculation bubble with the Reynolds number, which ranges from a tiny size ($l_{rec}/D = 0.04$) at $Re = 1$ to the maximum achievable value, which is the interbaffle spacing, at $Re = 70$.

6. Thermal-hydraulic results

6.1. Pressure drop

In this subsection, the main pressure drop results are presented for Reynolds numbers from $Re = 7$ to $Re = 3000$, and the different flow regions are



(a) $Re = 1$.

(b) $Re = 15$.

(c) $Re = 32$.

(d) $Re = 105$.

Figure 6: CFD contour of non-dimensional velocity, v/v_{med} in the symmetry plane of the OBR.

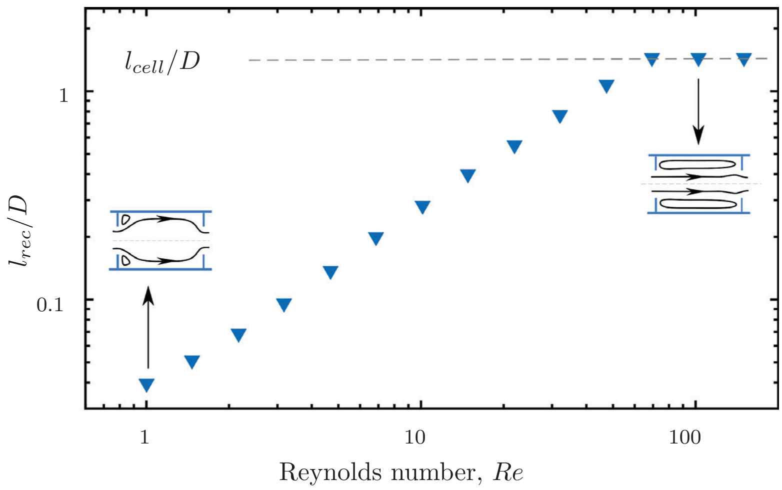


Figure 7: Recirculation zone size as a function of the Reynolds number.

identified. Fig. 8 (a) shows experimental results of the Fanning friction factor versus the Reynolds number for the geometry under study. The analytical solution for a smooth tube in laminar regime and the result obtained for an orifice plate, with an orifice coefficient equals to 0.8, are represented as well. The standard deviation of the pressure drop signal, sampled at a frequency $f = 2$ Hz, is also presented in Fig. 8 (b). The analysis of both results allow to distinguish three different flow regions:

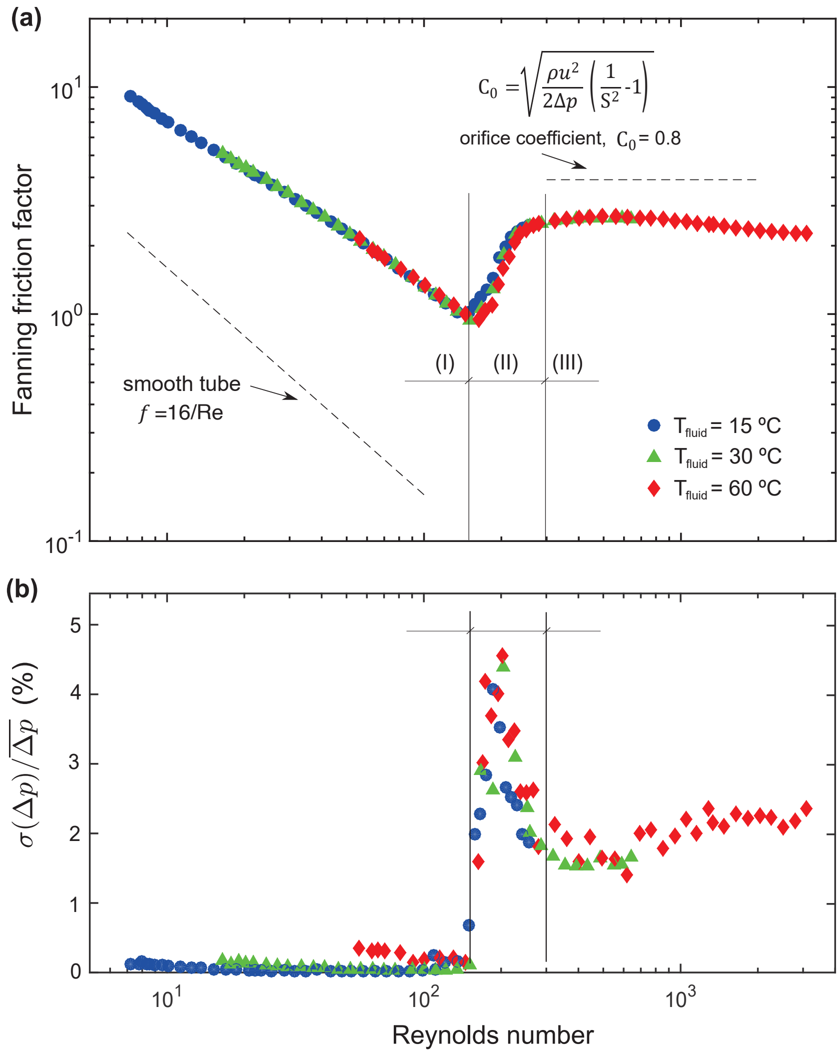


Figure 8: Fanning friction factor as a function of the Reynolds number for several tested fluid temperatures.

351 Region I: $Re < 160$

352 In this region, the flow presents characteristics of laminar flow, with a strong
353 influence of the Reynolds number on the friction factor, what agrees with

the visualization results showed in Section 4. Likewise, the low fluctuation of the pressure drop signal -which is the lowest among all the regions- is a good measure of the steadiness of the flow. Besides, friction factor augmentations of about 8 times, compared to the smooth tube, are identified. Pressure drop in this region can be estimated by the following correlation, which matches the experimental results with an accuracy of 10%:

$$f = 36.5\ Re^{-0.709} \qquad\qquad 7 < Re < 160 \tag{5}$$

Region II: $160 < Re < 300$

An abrupt change in friction factor in this region indicates the onset of transition to turbulence. Intense fluctuations of the pressure drop signal are identified in Fig. 8 (b), which correspond to the unstable behavior that has also been detected in the flow visualization tests for the same operational regime (see Figs. 4c and 5b, and the corresponding videos).

Region III: $Re > 300$

The turbulent flow regime is achieved from $Re \approx 300$ onwards. The standard deviation of the pressure signal is stabilized, with higher values than those of the laminar flow regime. The low influence of the Reynolds number on friction factor is also noticeable. The following correlation is proposed for the range $300 < Re < 3000$, which meets all the experimental results with an accuracy of 8%:

$$f = 3.14\ Re^{-0.033} \qquad\qquad 300 < Re < 3000 \tag{6}$$

As reflected in Fig. 8 (a), the friction factor is much lower than the one expected for an orifice with the same open area ratio. This can be justified by the interaction between consecutive baffles, after crossing a baffle the flow is not developed before reaching the next one, so the effect can not be described as a series of independent baffles.

6.2. Heat transfer

In this subsection, the main heat transfer results are provided for Reynolds numbers from 10 to 2200, and Prandtl numbers from 150 to 630. Fig. 9 shows experimental results of the Nusselt number as a function of the Reynolds number for three Prandtl numbers, $Pr = 150$, 285 and 630. Based on the influence of Reynolds number, three different flow regions can be clearly distinguished: laminar, transitional and turbulent. In the laminar region ($Re < 160$) there is a low influence of the Reynolds number on heat transfer coefficients are significantly higher than those that would be obtained for a smooth tube working under mixed convection conditions. The Reynolds number influence is quite sharp in the transitional region ($160 < Re < 300$). Onwards, the onset of turbulence highly increases the heat transfer rates. This outbreak occurs for $Re > 300$, which proves the suitability of the insert baffles as turbulence promoters for heat transfer increment purposes.

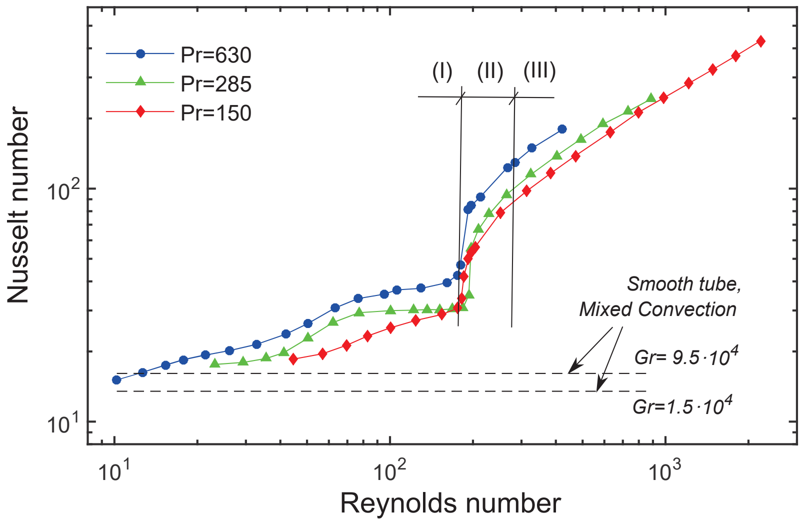


Figure 9: Nusselt number vs Reynolds number for three Prandtl numbers.

These data can be adjusted to correlations which as well show the influence of the Reynolds and Prandtl numbers on the Nusselt number. Eq. 7 corresponds to region I (laminar) and Eq. 8 to region III (turbulent).

$$Nu = 1.304\ Re^{0.402}\ Pr^{0.227} \qquad\qquad 10 < Re < 100 \tag{7}$$

$$Nu = 0.503\ Re^{0.735}\ Pr^{0.216} \qquad\qquad 300 < Re < 2200 \tag{8}$$

These equations are suitable for the Prandtl number range $130 < Pr < 650$. The exponent of the Prandtl number is similar for both regions (considering the experimental error). The fitting is quite satisfactory for both regions, with all the experimental results included in a range of +-10 % of the correlation, as can be seen in Fig. 10.

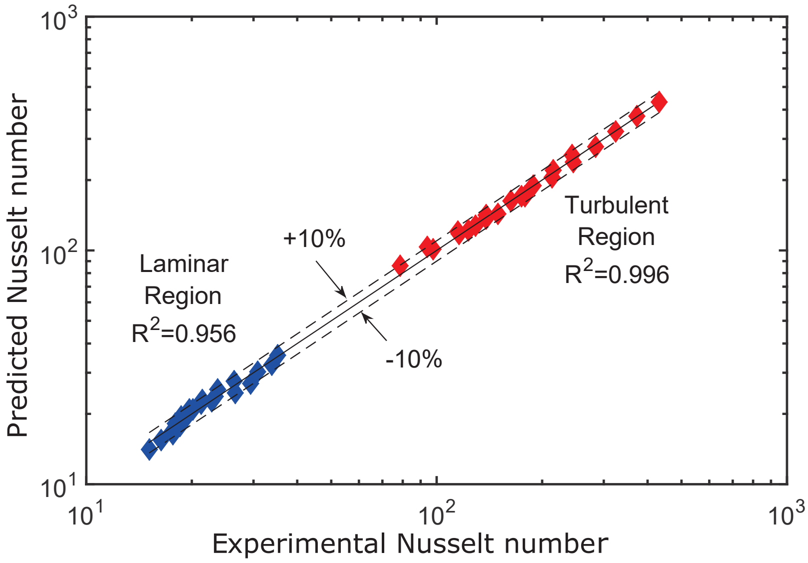


Figure 10: Comparison of the experimental Nu number and the predicted Nu number using Eq. 7 and 8.

A different way to identify the end of the laminar flow regime is using the standard deviation of the wall temperature [25]. Fig. 11 shows the mean standard deviation of all the measured wall temperatures as a function of the Reynolds number. For low Reynolds numbers the standard deviation is low and corresponds to the measurement noise. A significant increase for $Re >$ 160 indicates the onset of the transitional regime, when the flow pulsation allows the colder central stream to contact the wall regions, increasing the wall temperature fluctuations. Above $Re = 300$, the value is higher than in the laminar region but stable again, pointing the onset of the turbulent regime.

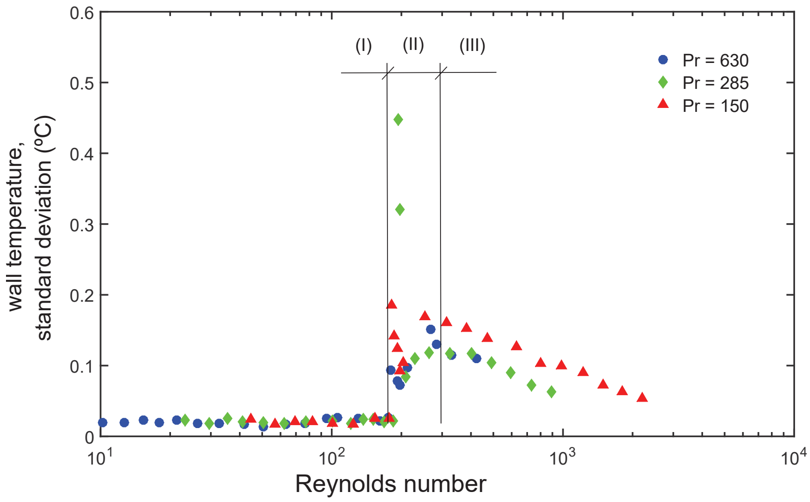


Figure 11: Standard deviation of the wall temperature vs Reynolds number.

Performance evaluation.

In order to evaluate the thermal enhancement of the geometry tested, the R3 criterion is used [26]. This parameter measures the heat transfer ratio between the tested geometry and an equivalent smooth tube for the same power consumption and basic geometry (number of tubes and tube diameter and length) (Eq. 9).

$$R3 = \frac{Nu}{Nu_s}\bigg|_{\dot{W}=\dot{W}_s, N=N_s, D=D_s, l=l_s} \tag{9}$$

For the same power consumption, well-known correlations of Nusselt number for the laminar [27] and turbulent [28] regions are used for calculating the term Nu_s. The results for $Pr = 150$ are represented in Fig. 12.

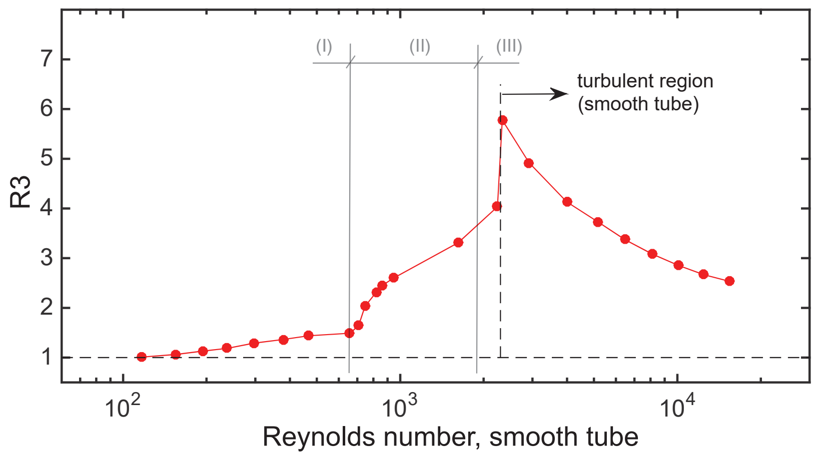


Figure 12: Performance evaluation criteria R3 vs smooth tube Reynolds number, $Pr = 150$.

As can be observed, the circular-orifice baffles show a poor performance for $Re_s < 700$, and a slight increase in the thermal performance with the Reynolds number. This is due to the increase in the Nusselt number in the laminar region when the Re is increased. Above $Re_s = 800$ there is a sharp rise in the thermal performance due to the onset of the transitional regime in the baffled tube. So, the range $800 < Re_s < 2300$ is where the highest performance can be expected. This trend is cut by the onset of the transitional regime in the smooth tube at $Re_s = 2300$, generating a descending trend in the performance. However, there is enhancement for all the range tested, with the lowest performance, R3 2.5, at $Re_s \approx 2 \times 10^4$.

7. Conclusions

1. An extensive study has been done in order to quantify the thermo-hydraulic performance of tubes with periodically-spaced circular-orifice

baffles. A wide range of dimensionless numbers has been studied ($Pr = 150 - 630$, $Re_n = 10 - 2200$).

2. Qualitative flow visualization tests show the flow patterns in the baffled tube in the three different flow regimes. In the laminar region, the size of the recirculation zones under the baffles characterizes the flow. Instability can be observed for $Re > 160$, this instability of the flow is related to the pulsation of the core stream. For a $Re > 300$ the flow can be described as chaotic.
3. Pressure drop and heat transfer measurements demonstrate the existence of three flow regions with distinguishable characteristics: a laminar region ($Re < 160$), a transitional region ($160 < Re < 300$) and a turbulent region ($Re > 300$).
4. The $R3$ performance criteria shows that the tested geometry can be advantageous, for $Pr = 150$, in comparison to smooth tubes working at a Reynolds number between 100 and $2 \cdot 10^4$.

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